

Implementation and Application of a new Plasticity Model in LS-DYNA including Lode Angle Dependence

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# OUTLINE

- 1. Motivation
  - Definition of the Lode angle and normalized third invariant
  - Materials with Lode angle dependent plasticity
  - Existing plasticity models with Lode angle dependency

## 2. New plasticity model

- Base assumptions and desired features
- Constitutive equations: yield function and flow rule
- Graphic representation on the  $\sigma_1$ - $\sigma_2$  space
- Numerical implementation

## 3. Numerical examples

- Tensile stress states: notched and flat grooved specimens
- Shear stress states: butterfly specimen
- Compressive stress states: upset test

## 4. Final remarks



#### Lode angle definition



Taken from "Xue, L. (2007),Ductile Fracture Modeling – Theory, Experimental Investigation, and Numerical Verification, PhD thesis, Massachusetts Institute Technology, Cambridge, MA." Geometrically, the Lode Angle is the "smallest angle between the line of pure shear and the projection of the stress tensor on the deviatoric plane" (L.Malcher et al.)

$$\theta_L = \tan^{-1} \left\{ \frac{1}{\sqrt{3}} \left[ 2 \left( \frac{s_2 - s_3}{s_1 - s_3} \right) - 1 \right] \right\}, \theta_L = \left[ 0, \frac{\pi}{3} \right]$$

• The Lode Angle  $\theta_L$  is relatated with the normalized third deviatoric stress invariant,  $\xi$ 

$$\xi = \cos(3\theta_L) = \frac{27}{2} \frac{\det(s)}{\sigma_{eq}^3}, \xi = [-1,1]$$

• Hereinafter, the normalized third deviatoric stress invariant,  $\xi$ , will be denoted as Lode Angle.

# In LS-DYNA, the normalized third deviatoric stress invariant, $\xi$ , is denoted as the "Lode Angle Parameter"



#### Lode Angle Definition, $\xi$



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#### **Aluminium alloys**

- Recent experimental analysis have proved that the Lode Angle have a considerable effect on the stress-plastic relation of some aluminium alloys (Y. Bai et al., Mirone et al, X. Gao et al.)
- In contrast, they also concluded that the hydrostatic pressure has a negligible effect





#### **Aluminium alloys**

- However, the effect of the Lode Angle is not constant.
- It depends on the stress state!

Stress State/Specimen	ξ	Lode Angle Effect (according to experimental evidences)
1. Tensile round bars	$\xi = 1$	Ν
2. Compression round bars	$\xi = -1$	N
3. Shear stress states	$\xi = 0$	Y
4. Plane Strain Specimen - Traction	$\xi = 0$	Y
5. Plane Strain Specimen – Compression	$\xi = 0$	Y
5. Bi-axial tension	$\xi = 1$	N
6. Bi-axial compression	$\xi = -1$	Ν

The maximum Lode Angle effect takes place when  $\xi = 0$ 



#### **Tresca's yield function**



$$\Phi = (\sigma_{max} - \sigma_{min}) - \sigma_y$$

#### Advantages:

• It is a classical, well-established yield function

- The effects of the 3rd invariant are fixed
- The yield function is non-continuous, making its numerical implementation more difficult (directional derivatives are needed!)



#### **Bai & Wierzbicki's yield function**



$$\Phi = \sigma_{eq} - \sigma_y [1 - c_{\Gamma}(\Gamma - \Gamma_0)] \left[ c_{\theta}^s + (c_{\theta}^{ax} - c_{\theta}^s) \left( \phi - \frac{\phi^{m+1}}{m+1} \right) \right]$$

#### Advantages:

• The convexity of the yield function has to be ensured by a numerical parameter

- 7 material parameters plus a hardening curve are required for calibration
- Isochoric plasticity is questionable
- Complex to implement



#### Hosford's yield function



$$\Phi = |\sigma_1 - \sigma_2|^m + |\sigma_2 - \sigma_3|^m + |\sigma_3 - \sigma_1|^m - 2\sigma_y^m$$

#### Advantages:

 It is available in LS-DYNA through MAT\_36 if parameters are set to be isotropic

- Effect of 3<sup>rd</sup> invariant can be controlled by the exponent "m", but the physical meaning of "m" is somewhat difficult to grasp
- Formulated in principal stress space



#### Gao's yield function



 $\Phi = c_1 (a_1 I_1^6 + 27 J_2^3 + b_1 J_3^2)^{1/6} - \sigma_{\gamma}$ 

 $a_1$  and  $b_1$  are material parameters and  $c_1$  is a function of  $a_1$  and  $b_1$ 

#### Advantages:

• It is possible to control the effect of the third invariant

- In addition to the hardening curve, 6 more material parameters are required which have no physical meaning
- Non-associated model
- The yield function may be non-convex!



# **2. New Plasticity Model**

### **Desired features**

- Starting from classical J<sub>2</sub> von Mises plasticity (\*MAT\_24)
- $J_2$  plasticity should be easily recovered if material is not dependent on  $\xi$
- Easiness of calibration (as less new parameters as possible)
- Dependency of  $\xi$  should be simple to grasp

#### Main assumptions for the new model

- Material has a different yield stress under tensile and shear stress states
- The dependency of the yield stress in respect to  $\xi$  is assumed quadratic



# **2. New Plasticity Model**

**Proposed yield function** 



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-200

-100

 $\overset{\circ}{\sigma}$ 

-300

-400

Parameters definition:

1. Hardening Curve: The same as used in von Mises Model

2. Tensile Yield Stress,  $\sigma_t$ . Determined through standard tensile test

3. Shear Yield Stress,  $\sigma_s$ . Determined through a shear test

400

500



# **2. New Plasticity Model**

#### Proposed yield function



#### Advantages:

- Only one extra parameter in comparison to classical J<sub>2</sub> plasticity (\*MAT\_24)
- ξ-dependency is easier to grasp than other plasticity models
- Only two physical tests are required for calibration: tensile and shear test
- The yield function is always continuous
- The model is relatively simple to implement

### **Disadvantage:**

• Yield surface may be non-convex!



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# **2. New Constitutive Model:** Constitutive relations

**Yield function** 

$$\Phi = \sigma_{eq} + (\sigma_t - \sigma_s). (1 - \xi^2) - \sigma_y = 0$$

#### **Associative plastic flow**

$$\dot{\boldsymbol{\varepsilon}}^{p} = \dot{\boldsymbol{\gamma}} \frac{\partial \Phi}{\partial \boldsymbol{\sigma}} = \left[ \sqrt{\frac{3}{2}} \frac{\boldsymbol{s}}{\|\boldsymbol{s}\|} - 2. \left(\sigma_{t} - \sigma_{s}\right) \boldsymbol{\xi} \frac{\partial \boldsymbol{\xi}}{\partial \boldsymbol{\sigma}} \right]$$

**Plastic work equivalence** 

$$\sigma_{eq}.\,\bar{\varepsilon}^p = \boldsymbol{\sigma}: \dot{\boldsymbol{\varepsilon}}^p$$



# **2. New Constitutive Model:** Numerical implementation

The following system of equations has to be solved:

$$res_{s} = s_{n+1} - 2G\varepsilon_{n+1}^{e\ Trial} + 2G\Delta\gamma \left[ \sqrt{\frac{3}{2}} \frac{s_{n+1}}{\|s_{n+1}\|} - 2.\left(\sigma_{t} - \sigma_{s}\right)\xi_{n+1} \frac{\partial\xi_{n+1}}{\partial\sigma_{n+1}} \right]$$
$$res_{\varepsilon^{p}} = \bar{\varepsilon}^{p}_{n+1} - \bar{\varepsilon}^{p}_{n} - \frac{\sigma_{n+1}:\Delta\varepsilon^{p}}{\sigma_{y}(\bar{\varepsilon}^{p}_{n+1})}$$
$$res_{\Delta\gamma} = \sigma_{eq_{n+1}} + (\sigma_{t} - \sigma_{s}).\left(1 - \xi_{n+1}^{2}\right) - \sigma_{y}(\bar{\varepsilon}^{p}_{n+1})$$

The consistent tangent operator (implicit analysis) reads:

$$\boldsymbol{D}^{ep} = \frac{\partial \boldsymbol{\sigma}_{n+1}}{\partial \boldsymbol{\varepsilon}_{n+1}^{e \ Trial}} = 2G[\boldsymbol{A}]^{-1} : \left(\boldsymbol{I}_4 - \frac{1}{3}\boldsymbol{I}_2 \otimes \boldsymbol{I}_2\right) + K\boldsymbol{I}_2 \otimes \boldsymbol{I}_2$$



Material Properties	
Young Modulus	$E = 71.15 \ GPa$
Poisson's Ration	v = 0.3
Tensile yield stress	$\sigma_t = 370 MPa$

#### **Tensile Stress States**









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**Tensile Stress States: Flat Grooved,** *R* = 1.59*mm* 80 The new model has the ability to properly 70 +++ capture the effects of  $\xi$ Reaction (kN) The appropriate value of  $\sigma_s = 325 MPa$ Despite the slight non-convexity of the yield • 20 von Mises function for  $\sigma_s = 325 MPa$ , no convergence -New Model, σ,=370, σ,=340 10 problems were found  $\rightarrow$  In fact, the New Model, σ,=370, σ,=325 convergence rate is practically quadratic 0.1 0.2 0.3 0.4 **Displacement (mm) Von Mises** New Model,  $\sigma_s = 325$ New Model,  $\sigma_s = 340$ EPBAR EPBAR EPBAR 0.094 0.188 0.0882 0.176 0.0867 0.173









#### **Shear Stress States: Butterfly Specimen**





**Compressive Stress State: Upset Test** 





- Friction coefficient:  $\mu = 0.05$
- D = 8mm, h = 6mm



- There is no significant difference between the results provided by the von Mises and the new model
- Results do not change with varying  $\sigma_s$
- The numerical and experimental results match perfectly



#### **Compressive Stress State: Upset Test**

#### Third Invariant, $\xi$





- Throughout the deformation process and at the center of the specimen,  $\xi$  remains practically constant and equal to 1
- However, on the outer surface,  $\xi$  is not constant, verifying a small decrease
- The evolution of  $\xi$  on the outer surface of the specimen does not have any impact on the final result for different values of  $\sigma_s$
- The results are in agreement with the main assumptions initially proposed



#### **Accumulated Plastic Strain**

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XΙ

-0.5

0

## **3. Numerical Results**

#### **Compressive Stress State: Butterfly Test**







## **Final Remarks**

- The proposed model is able to capture the effects of the third invariant at different stress states (tension, shear and compression)
- Only one extra material parameter is required by the proposed model when compared to classical von Mises plasticity; the effects of the additional material parameter are also easy to grasp
- From the computational point of view, the proposed model is simple to implement and also to understand; furthermore, it does not present convergence problems
- The initial desired features for the model were achieved



## **Final Remarks**

## **Future work**

- To perform an analytical analysis to find out the critical value of  $\sigma_t \sigma_s$  from which the resulting yield function is non-convex
- To define based on experimental tests how to define  $\sigma_s$  (flat grooved specimen, pure shear test, ...)
- To combine the new model with a damage/failure model (such as GISSMO through \*MAT\_ADD\_EROSION)
- To use the new model to simulate components and structures in practical applications
- To better investigate the yield locus of aluminium alloys based on experimental evidence (is non-convexity perhaps a must in some cases?!)



## **Thank You!**

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